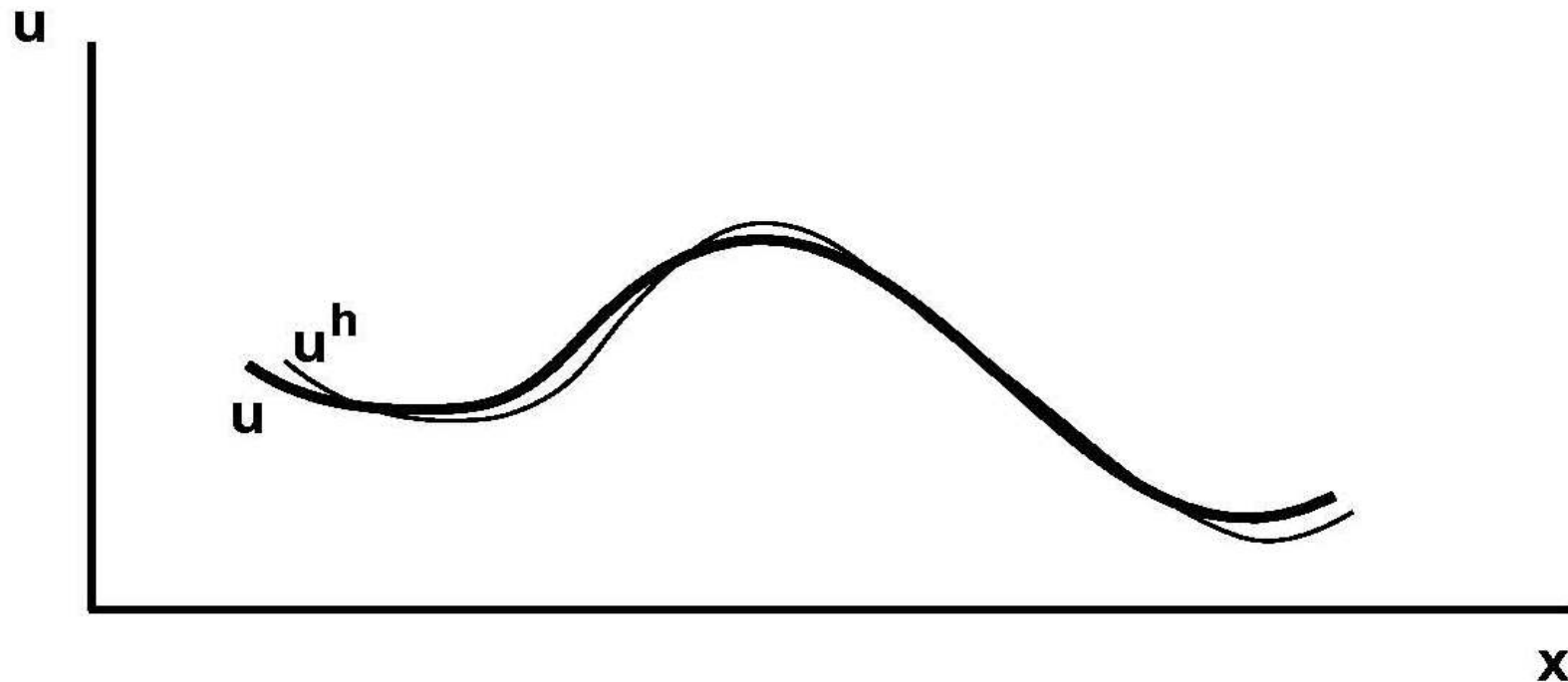




Approximation Theory

To do: given $u(x)$ in Ω , approximate by known functions

$$u(x) \approx u^h(x) = f^i(x)a_i = N^i(x)\hat{u}_i$$



Approximation of Functions



Least Squares Problem

$$\begin{aligned} I_{ls} &= \int_{\Omega} (\epsilon^h)^2 d\Omega = \int_{\Omega} (u^h - u)^2 d\Omega \\ &= \int_{\Omega} (N^k a_k - u)^2 d\Omega \rightarrow \min \end{aligned}$$

\Rightarrow

$$\delta I_{ls} = \delta a_k \int_{\Omega} N^k (N^l a_l - u) d\Omega = 0$$

$$\int_{\Omega} N^k N^l d\Omega a_l = \int_{\Omega} N^k u d\Omega$$

\Rightarrow Equivalent to Galerkin WRM

\Rightarrow Choice of W^i from same set as N^i optimal



2. Weighted Residual Methods(WRM):

Define: $\epsilon^h = u - u^h$ - the error or residual

Require: $\epsilon^h \rightarrow 0$ in Ω

Introduce a set of weighting functions W^i ; $i = 1, 2, \dots, M$

Require that:

$$\int_{\Omega} W^i \epsilon^h d\Omega = 0 \quad , \quad i = 1, 2, \dots, M$$

Then, as $M \rightarrow \infty$, $\epsilon^h \rightarrow 0$ at all points in Ω



FROM APPROXIMATION TO OPERATORS

Before: given u , approximate: $\|u - u^h\| \rightarrow \min$

Now: given $L(u) = 0$ approximate: $\|L(u) - L(u^h)\| \rightarrow \min$

$\Rightarrow \|L(u^h)\| \rightarrow \min$

Minimize error:

$$\epsilon_L^h = L(u^h) = L(N^i \hat{u}_i)$$

using WRM

$$\int_{\Omega} W^i \epsilon_L^h d\Omega = 0, \quad i = 1, M$$



From Approximation to Operators

Choice of N^i, W^i defines the method:

- N^i polynomial, $W^i = \delta(x_i)$: FDM
- N^i polynomial, $W^i = 1$ if $x \in \Omega_{el}$, 0 otherwise : FVM
- N^i polynomial, $W^i = N^i$: GFEM
- N^i polynomial, $W^i \neq N^i$: Petrov-GFEM
- N^i spectral, $W^i = \delta(x_i)$: SEM



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Description of Governing Equations



CFD Notations

PDE of p -th order $f\left(u, \mathbf{x}, t, \frac{\partial u}{\partial x_1}, \dots, \frac{\partial u}{\partial x_n}, \frac{\partial u}{\partial t}, \frac{\partial^2 u}{\partial x_1 \partial x_2}, \dots, \frac{\partial^p u}{\partial t^p}\right) = 0$

scalar unknowns $u = u(\mathbf{x}, t), \quad \mathbf{x} \in \mathbb{R}^n, \quad t \in \mathbb{R}, \quad n = 1, 2, 3$

vector unknowns $\mathbf{v} = \mathbf{v}(\mathbf{x}, t), \quad \mathbf{v} \in \mathbb{R}^m, \quad m = 1, 2, \dots$

Nabla operator

$$\nabla = \mathbf{i} \frac{\partial}{\partial x} + \mathbf{j} \frac{\partial}{\partial y} + \mathbf{k} \frac{\partial}{\partial z}$$

$$\mathbf{x} = (x, y, z), \quad \mathbf{v} = (v_x, v_y, v_z)$$

$$\nabla u = \mathbf{i} \frac{\partial u}{\partial x} + \mathbf{j} \frac{\partial u}{\partial y} + \mathbf{k} \frac{\partial u}{\partial z} = \left[\frac{\partial u}{\partial x}, \frac{\partial u}{\partial y}, \frac{\partial u}{\partial z} \right]^T$$

gradient

$$\nabla \cdot \mathbf{v} = \frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z}$$

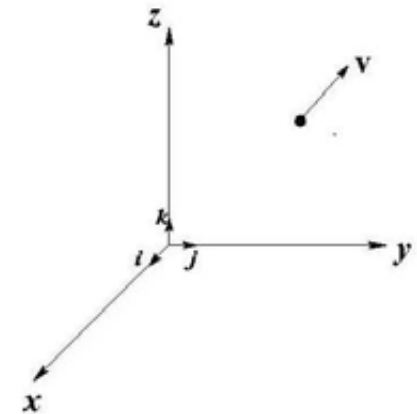
divergence

$$\nabla \times \mathbf{v} = \det \begin{bmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ v_x & v_y & v_z \end{bmatrix} = \begin{bmatrix} \frac{\partial v_z}{\partial y} - \frac{\partial v_y}{\partial z} \\ \frac{\partial v_x}{\partial z} - \frac{\partial v_z}{\partial x} \\ \frac{\partial v_y}{\partial x} - \frac{\partial v_x}{\partial y} \end{bmatrix}$$

curl

$$\Delta u = \nabla \cdot (\nabla u) = \nabla^2 u = \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 u}{\partial z^2}$$

Laplacian

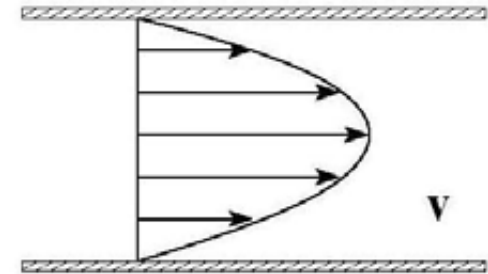




CFD Notations

Velocity gradient

$$\nabla \mathbf{v} = [\nabla v_x, \nabla v_y, \nabla v_z] = \begin{bmatrix} \frac{\partial v_x}{\partial x} & \frac{\partial v_x}{\partial y} & \frac{\partial v_x}{\partial z} \\ \frac{\partial v_y}{\partial x} & \frac{\partial v_y}{\partial y} & \frac{\partial v_y}{\partial z} \\ \frac{\partial v_z}{\partial x} & \frac{\partial v_z}{\partial y} & \frac{\partial v_z}{\partial z} \end{bmatrix}$$



Remark. The trace (sum of diagonal elements) of $\nabla \mathbf{v}$ equals $\nabla \cdot \mathbf{v}$.

Deformation rate tensor (symmetric part of $\nabla \mathbf{v}$)

$$\mathcal{D}(\mathbf{v}) = \frac{1}{2}(\nabla \mathbf{v} + \nabla \mathbf{v}^T) = \begin{bmatrix} \frac{\partial v_x}{\partial x} & \frac{1}{2} \left(\frac{\partial v_y}{\partial x} + \frac{\partial v_x}{\partial y} \right) & \frac{1}{2} \left(\frac{\partial v_z}{\partial x} + \frac{\partial v_x}{\partial z} \right) \\ \frac{1}{2} \left(\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right) & \frac{\partial v_y}{\partial y} & \frac{1}{2} \left(\frac{\partial v_z}{\partial y} + \frac{\partial v_y}{\partial z} \right) \\ \frac{1}{2} \left(\frac{\partial v_x}{\partial z} + \frac{\partial v_z}{\partial x} \right) & \frac{1}{2} \left(\frac{\partial v_y}{\partial z} + \frac{\partial v_z}{\partial y} \right) & \frac{\partial v_z}{\partial z} \end{bmatrix}$$

Spin tensor $\mathcal{S}(\mathbf{v}) = \nabla \mathbf{v} - \mathcal{D}(\mathbf{v})$ (skew-symmetric part of $\nabla \mathbf{v}$)



CFD Notations

Scalar product of two vectors

$$\mathbf{a}, \mathbf{b} \in \mathbb{R}^3, \quad \mathbf{a} \cdot \mathbf{b} = \mathbf{a}^T \mathbf{b} = [a_1 \ a_2 \ a_3] \begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix} = a_1 b_1 + a_2 b_2 + a_3 b_3 \in \mathbb{R}$$

Example. $\mathbf{v} \cdot \nabla u = v_x \frac{\partial u}{\partial x} + v_y \frac{\partial u}{\partial y} + v_z \frac{\partial u}{\partial z}$ *convective derivative*

Dyadic product of two vectors

$$\mathbf{a}, \mathbf{b} \in \mathbb{R}^3, \quad \mathbf{a} \otimes \mathbf{b} = \mathbf{a} \mathbf{b}^T = \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} [b_1 \ b_2 \ b_3] = \begin{bmatrix} a_1 b_1 & a_1 b_2 & a_1 b_3 \\ a_2 b_1 & a_2 b_2 & a_2 b_3 \\ a_3 b_1 & a_3 b_2 & a_3 b_3 \end{bmatrix} \in \mathbb{R}^{3 \times 3}$$



CFD Notations

$$1. \quad \alpha \mathcal{T} = \{\alpha t_{ij}\}, \quad \mathcal{T} = \{t_{ij}\} \in \mathbb{R}^{3 \times 3}, \quad \alpha \in \mathbb{R}$$

$$2. \quad \mathcal{T}^1 + \mathcal{T}^2 = \{t_{ij}^1 + t_{ij}^2\}, \quad \mathcal{T}^1, \mathcal{T}^2 \in \mathbb{R}^{3 \times 3}, \quad \mathbf{a} \in \mathbb{R}^3$$

$$3. \quad \mathbf{a} \cdot \mathcal{T} = [a_1, a_2, a_3] \begin{bmatrix} t_{11} & t_{12} & t_{13} \\ t_{21} & t_{22} & t_{23} \\ t_{31} & t_{32} & t_{33} \end{bmatrix} = \sum_{i=1}^3 a_i \underbrace{[t_{i1}, t_{i2}, t_{i3}]}_{i\text{-th row}}$$

$$4. \quad \mathcal{T} \cdot \mathbf{a} = \begin{bmatrix} t_{11} & t_{12} & t_{13} \\ t_{21} & t_{22} & t_{23} \\ t_{31} & t_{32} & t_{33} \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} = \sum_{j=1}^3 \begin{bmatrix} t_{1j} \\ t_{2j} \\ t_{3j} \end{bmatrix} a_j \quad (j\text{-th column})$$

$$5. \quad \mathcal{T}^1 \cdot \mathcal{T}^2 = \begin{bmatrix} t_{11}^1 & t_{12}^1 & t_{13}^1 \\ t_{21}^1 & t_{22}^1 & t_{23}^1 \\ t_{31}^1 & t_{32}^1 & t_{33}^1 \end{bmatrix} \begin{bmatrix} t_{11}^2 & t_{12}^2 & t_{13}^2 \\ t_{21}^2 & t_{22}^2 & t_{23}^2 \\ t_{31}^2 & t_{32}^2 & t_{33}^2 \end{bmatrix} = \left\{ \sum_{k=1}^3 t_{ik}^1 t_{kj}^2 \right\}$$

$$6. \quad \mathcal{T}^1 : \mathcal{T}^2 = \text{tr}(\mathcal{T}^1 \cdot (\mathcal{T}^2)^T) = \sum_{i=1}^3 \sum_{k=1}^3 t_{ik}^1 t_{ik}^2$$



Conservation Laws

Quantities Conserved:

- Mass
- Momentum (Newton' Law)
- Energy (First Law of Thermodynamics)

General Form:

$$Change = Production(Destruction)$$

In Eulerian Frame:

$$Change|_{fixed\mathbf{x}} + Transport = Diffusion + Production$$

Or:

$$\Phi_{,t} + \nabla \cdot \mathbf{v}\Phi = \nabla \cdot \mathbf{q} + S$$



Conservation Laws

Physical principles

1. Mass is conserved
2. Newton's second law
3. Energy is conserved



Mathematical equations

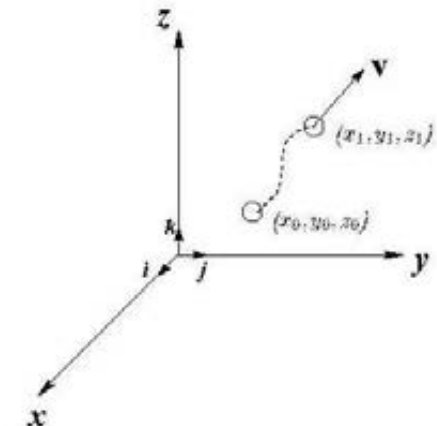
- continuity equation
- momentum equations
- energy equation

It is important to understand the meaning and significance of each equation in order to develop a good numerical method and properly interpret the results

Description of fluid motion

Eulerian monitor the flow characteristics
in a fixed control volume

Lagrangian track individual fluid particles as
they move through the flow field



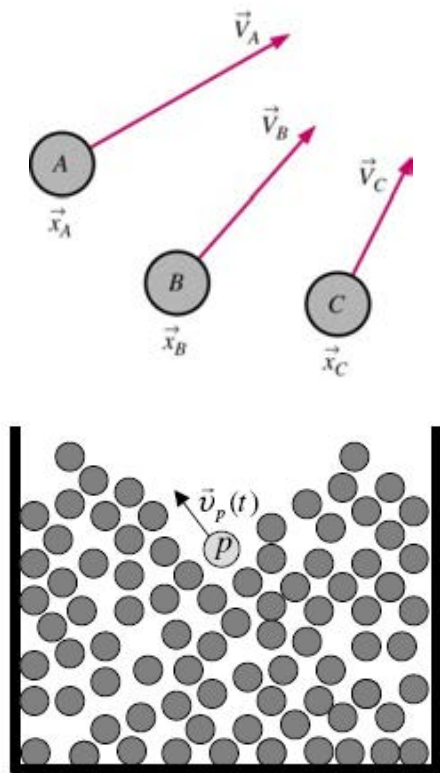


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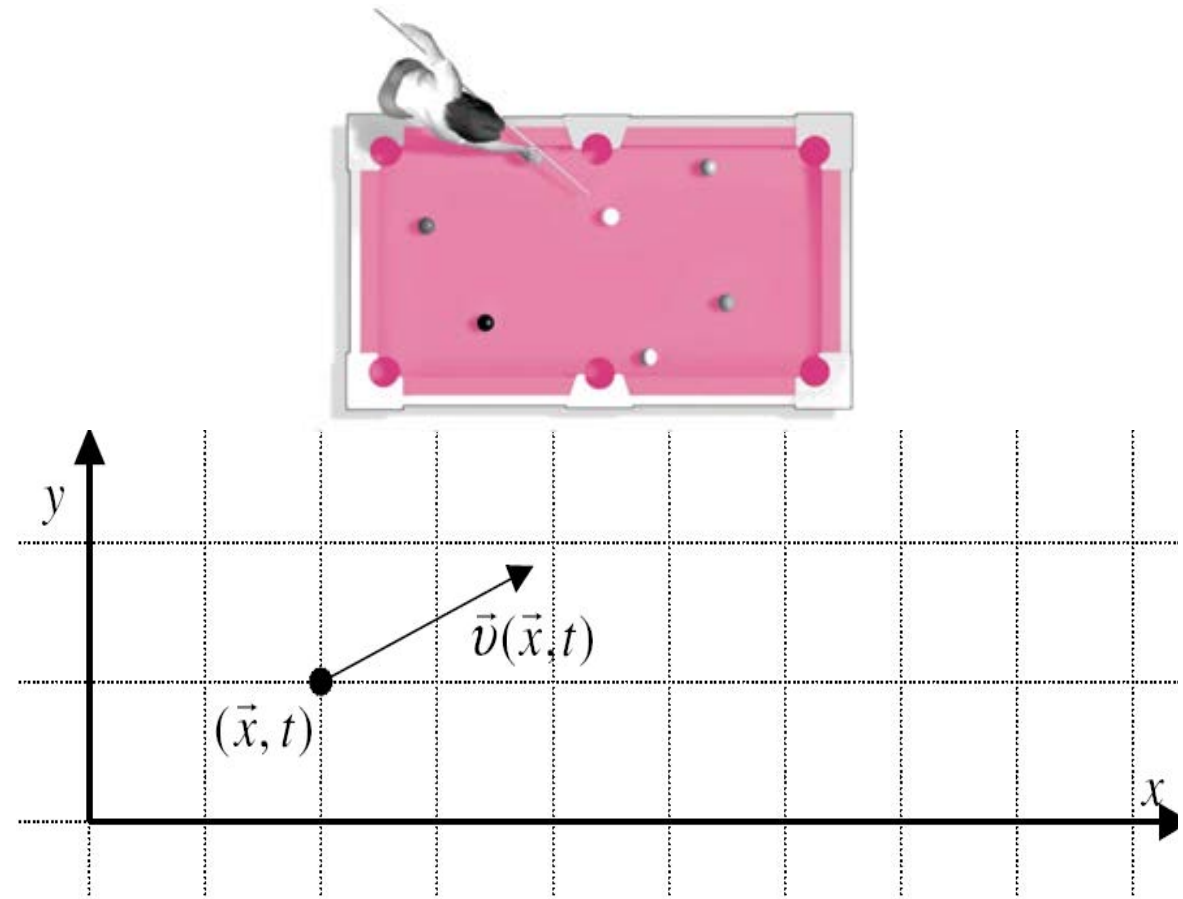
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Two ways of describing a fluid flow

Lagrangian description, Eulerian description



Lagrangian description; snapshot



Eulerian description; Cartesian grid

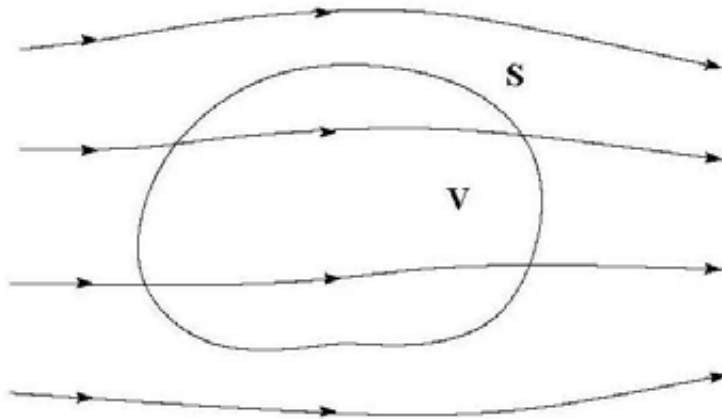


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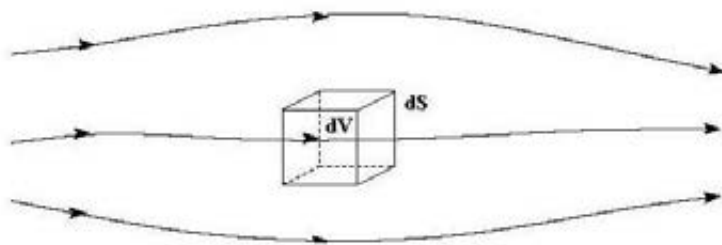
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Flow models and reference frames

Eulerian

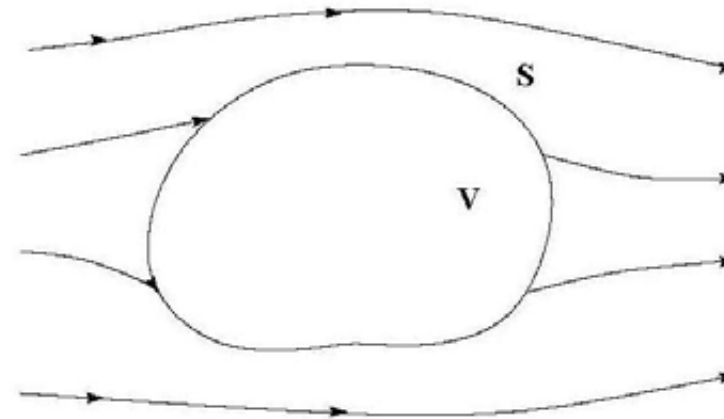


fixed CV of a finite size

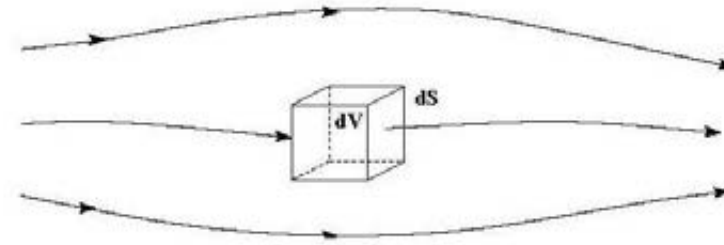


fixed infinitesimal CV

Lagrangian



moving CV of a finite size



moving infinitesimal CV

integral

differential

Good news: all flow models lead to the same equations



Eulerian vs. Lagrangian

Definition. Substantial time derivative $\frac{d}{dt}$ is the rate of change for a moving fluid particle. Local time derivative $\frac{\partial}{\partial t}$ is the rate of change at a fixed point.

Let $u = u(\mathbf{x}, t)$, where $\mathbf{x} = \mathbf{x}(\mathbf{x}_0, t)$. The chain rule yields

$$\frac{du}{dt} = \frac{\partial u}{\partial t} + \frac{\partial u}{\partial x} \frac{dx}{dt} + \frac{\partial u}{\partial y} \frac{dy}{dt} + \frac{\partial u}{\partial z} \frac{dz}{dt} = \frac{\partial u}{\partial t} + \mathbf{v} \cdot \nabla u$$

substantial derivative = local derivative + convective derivative

Reynolds transport theorem

$$\frac{d}{dt} \int_{V_t} u(\mathbf{x}, t) dV = \int_{V \equiv V_t} \frac{\partial u(\mathbf{x}, t)}{\partial t} dV + \int_{S \equiv S_t} u(\mathbf{x}, t) \mathbf{v} \cdot \mathbf{n} dS$$

rate of change in a moving volume = rate of change in a fixed volume + convective transfer through the surface



$$\underbrace{\frac{D}{Dt}}_{\text{Lagrangian}} \equiv \underbrace{\frac{\partial}{\partial t} + \vec{v}_p \cdot \nabla}_{\text{Eulerian}}$$

$$\underbrace{\frac{D\vec{G}}{Dt}}_{\text{Lagrangian rate of change}} = \underbrace{\frac{\partial \vec{G}}{\partial t}}_{\text{Eulerian rate of change}} + \underbrace{\vec{v} \cdot \nabla \vec{G}}_{\text{Convective rate of change}}$$

$$\underbrace{\frac{D\vec{v}}{Dt}}_{\text{Lagrangian acceleration}} = \underbrace{\frac{\partial \vec{v}}{\partial t}}_{\text{Eulerian acceleration}} + \underbrace{\vec{v} \cdot \nabla \vec{v}}_{\text{Convective acceleration}}$$



Continuity Equation

Physical principle: conservation of mass

$$\frac{dm}{dt} = \frac{d}{dt} \int_{V_t} \rho dV = \int_{V \equiv V_t} \frac{\partial \rho}{\partial t} dV + \int_{S \equiv S_t} \rho \mathbf{v} \cdot \mathbf{n} dS = 0$$

accumulation of mass inside CV = net influx through the surface

Divergence theorem yields

Continuity equation

$$\int_V \left[\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) \right] dV = 0 \quad \Rightarrow \quad \boxed{\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0}$$

Lagrangian representation

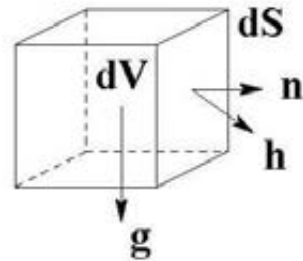
$$\nabla \cdot (\rho \mathbf{v}) = \mathbf{v} \cdot \nabla \rho + \rho \nabla \cdot \mathbf{v} \quad \Rightarrow \quad \frac{d\rho}{dt} + \rho \nabla \cdot \mathbf{v} = 0$$

Incompressible flows: $\frac{d\rho}{dt} = \nabla \cdot \mathbf{v} = 0$ (constant density)



Momentum Equation

Physical principle: $\mathbf{f} = m\mathbf{a}$ (Newton's second law)



total force $\mathbf{f} = \rho \mathbf{g} dV + \mathbf{h} dS$, where $\mathbf{h} = \boldsymbol{\sigma} \cdot \mathbf{n}$
 body forces \mathbf{g} gravitational, electromagnetic, ...
 surface forces \mathbf{h} pressure + viscous stress

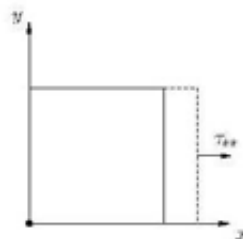
Stress tensor $\boldsymbol{\sigma} = -p\mathbf{I} + \boldsymbol{\tau}$ momentum flux

For a *newtonian fluid* viscous stress is proportional to velocity gradients:

$$\boldsymbol{\tau} = (\lambda \nabla \cdot \mathbf{v})\mathbf{I} + 2\mu \mathcal{D}(\mathbf{v}), \quad \text{where} \quad \mathcal{D}(\mathbf{v}) = \frac{1}{2}(\nabla \mathbf{v} + \nabla \mathbf{v}^T), \quad \lambda \approx -\frac{2}{3}\mu$$

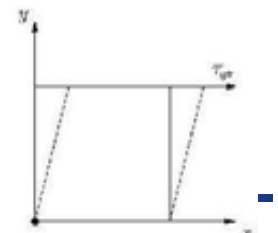
Normal stress: *stretching*

$$\begin{aligned} \tau_{xx} &= \lambda \nabla \cdot \mathbf{v} + 2\mu \frac{\partial v_x}{\partial x} \\ \tau_{yy} &= \lambda \nabla \cdot \mathbf{v} + 2\mu \frac{\partial v_y}{\partial y} \\ \tau_{zz} &= \lambda \nabla \cdot \mathbf{v} + 2\mu \frac{\partial v_z}{\partial z} \end{aligned}$$



Shear stress: *deformation*

$$\begin{aligned} \tau_{xy} &= \tau_{yx} = \mu \left(\frac{\partial v_y}{\partial x} + \frac{\partial v_x}{\partial y} \right) \\ \tau_{xz} &= \tau_{zx} = \mu \left(\frac{\partial v_x}{\partial z} + \frac{\partial v_z}{\partial x} \right) \\ \tau_{yz} &= \tau_{zy} = \mu \left(\frac{\partial v_z}{\partial y} + \frac{\partial v_y}{\partial z} \right) \end{aligned}$$





Momentum Equation

Newton's law for a moving volume

$$\begin{aligned}
\frac{d}{dt} \int_{V_t} \rho \mathbf{v} dV &= \int_{V \equiv V_t} \frac{\partial(\rho \mathbf{v})}{\partial t} dV + \int_{S \equiv S_t} (\rho \mathbf{v} \otimes \mathbf{v}) \cdot \mathbf{n} dS \\
&= \int_{V \equiv V_t} \rho \mathbf{g} dV + \int_{S \equiv S_t} \boldsymbol{\sigma} \cdot \mathbf{n} dS
\end{aligned}$$

Transformation of surface integrals

$$\int_V \left[\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) \right] dV = \int_V [\nabla \cdot \boldsymbol{\sigma} + \rho \mathbf{g}] dV, \quad \boldsymbol{\sigma} = -p\mathbf{I} + \boldsymbol{\tau}$$

Momentum equations

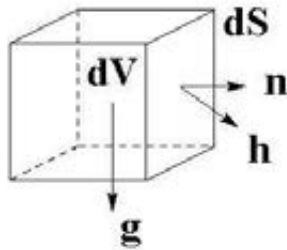
$$\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = -\nabla p + \nabla \cdot \boldsymbol{\tau} + \rho \mathbf{g}$$

$$\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = \rho \underbrace{\left[\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} \right]}_{\text{substantial derivative}} + \mathbf{v} \underbrace{\left[\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) \right]}_{\text{continuity equation}} = \rho \frac{d\mathbf{v}}{dt}$$



Energy Equation

Physical principle: $\delta e = s + w$ (first law of thermodynamics)



δe accumulation of internal energy

s heat transmitted to the fluid particle

w rate of work done by external forces

Heating: $s = \rho q dV - f_q dS$

q internal heat sources

f_q diffusive heat transfer

T absolute temperature

κ thermal conductivity

Fourier's law of heat conduction

$$f_q = -\kappa \nabla T$$

the heat flux is proportional to the local temperature gradient

Work done per unit time = total force \times velocity

$$w = \mathbf{f} \cdot \mathbf{v} = \rho \mathbf{g} \cdot \mathbf{v} dV + \mathbf{v} \cdot (\boldsymbol{\sigma} \cdot \mathbf{n}) dS, \quad \boldsymbol{\sigma} = -p\mathbf{I} + \boldsymbol{\tau}$$



Energy Equation

Total energy per unit mass: $E = e + \frac{|\mathbf{v}|^2}{2}$

e specific internal energy due to random molecular motion

$\frac{|\mathbf{v}|^2}{2}$ specific kinetic energy due to translational motion

Integral conservation law for a moving volume

$$\begin{aligned}
\frac{d}{dt} \int_{V_t} \rho E dV &= \int_{V \equiv V_t} \frac{\partial(\rho E)}{\partial t} dV + \int_{S \equiv S_t} \rho E \mathbf{v} \cdot \mathbf{n} dS && \text{accumulation} \\
&= \int_{V \equiv V_t} \rho q dV + \int_{S \equiv S_t} \kappa \nabla T \cdot \mathbf{n} dS && \text{heating} \\
&+ \int_{V \equiv V_t} \rho \mathbf{g} \cdot \mathbf{v} dV + \int_{S \equiv S_t} \mathbf{v} \cdot (\boldsymbol{\sigma} \cdot \mathbf{n}) dS && \text{work done}
\end{aligned}$$

Transformation of surface integrals

$$\int_V \left[\frac{\partial(\rho E)}{\partial t} + \nabla \cdot (\rho E \mathbf{v}) \right] dV = \int_V [\nabla \cdot (\kappa \nabla T) + \rho q + \nabla \cdot (\boldsymbol{\sigma} \cdot \mathbf{v}) + \rho \mathbf{g} \cdot \mathbf{v}] dV,$$

where $\nabla \cdot (\boldsymbol{\sigma} \cdot \mathbf{v}) = -\nabla \cdot (p\mathbf{v}) + \nabla \cdot (\boldsymbol{\tau} \cdot \mathbf{v}) = -\nabla \cdot (p\mathbf{v}) + \mathbf{v} \cdot (\nabla \cdot \boldsymbol{\tau}) + \nabla \mathbf{v} : \boldsymbol{\tau}$



Energy Equation

Total energy equation

$$\frac{\partial(\rho E)}{\partial t} + \nabla \cdot (\rho E \mathbf{v}) = \nabla \cdot (\kappa \nabla T) + \rho q - \nabla \cdot (p \mathbf{v}) + \mathbf{v} \cdot (\nabla \cdot \boldsymbol{\tau}) + \nabla \mathbf{v} : \boldsymbol{\tau} + \rho \mathbf{g} \cdot \mathbf{v}$$

$$\frac{\partial(\rho E)}{\partial t} + \nabla \cdot (\rho E \mathbf{v}) = \underbrace{\rho \left[\frac{\partial E}{\partial t} + \mathbf{v} \cdot \nabla E \right]}_{\text{substantial derivative}} + \underbrace{E \left[\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) \right]}_{\text{continuity equation}} = \rho \frac{dE}{dt}$$

Momentum equations $\rho \frac{d\mathbf{v}}{dt} = -\nabla p + \nabla \cdot \boldsymbol{\tau} + \rho \mathbf{g}$ (Lagrangian form)

$$\rho \frac{dE}{dt} = \rho \frac{de}{dt} + \mathbf{v} \cdot \rho \frac{d\mathbf{v}}{dt} = \frac{\partial(\rho e)}{\partial t} + \nabla \cdot (\rho e \mathbf{v}) + \mathbf{v} \cdot [-\nabla p + \nabla \cdot \boldsymbol{\tau} + \rho \mathbf{g}]$$

Internal energy equation

$$\frac{\partial(\rho e)}{\partial t} + \nabla \cdot (\rho e \mathbf{v}) = \nabla \cdot (\kappa \nabla T) + \rho q - p \nabla \cdot \mathbf{v} + \nabla \mathbf{v} : \boldsymbol{\tau}$$



Summary of GE

1. Continuity equation / conservation of mass

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$$

2. Momentum equations / Newton's second law

$$\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = -\nabla p + \nabla \cdot \tau + \rho \mathbf{g}$$

3. Energy equation / first law of thermodynamics

$$\frac{\partial(\rho E)}{\partial t} + \nabla \cdot (\rho E \mathbf{v}) = \nabla \cdot (\kappa \nabla T) + \rho q - \nabla \cdot (p \mathbf{v}) + \mathbf{v} \cdot (\nabla \cdot \tau) + \nabla \mathbf{v} : \tau + \rho \mathbf{g} \cdot \mathbf{v}$$

$$E = e + \frac{|\mathbf{v}|^2}{2}, \quad \frac{\partial(\rho e)}{\partial t} + \nabla \cdot (\rho e \mathbf{v}) = \nabla \cdot (\kappa \nabla T) + \rho q - p \nabla \cdot \mathbf{v} + \nabla \mathbf{v} : \tau$$

This PDE system is referred to as the *compressible Navier-Stokes equations*



Conservation form of GE

Generic conservation law for a scalar quantity

$$\frac{\partial u}{\partial t} + \nabla \cdot \mathbf{f} = q, \quad \text{where } \mathbf{f} = \mathbf{f}(u, \mathbf{x}, t) \text{ is the flux function}$$

Conservative variables, fluxes and sources

$$U = \begin{bmatrix} \rho \\ \rho \mathbf{v} \\ \rho E \end{bmatrix}, \quad \mathbf{F} = \begin{bmatrix} \rho \mathbf{v} \\ \rho \mathbf{v} \otimes \mathbf{v} + p \mathbf{I} - \boldsymbol{\tau} \\ (\rho E + p) \mathbf{v} - \kappa \nabla T - \boldsymbol{\tau} \cdot \mathbf{v} \end{bmatrix}, \quad Q = \begin{bmatrix} 0 \\ \rho \mathbf{g} \\ \rho(q + \mathbf{g} \cdot \mathbf{v}) \end{bmatrix}$$

Navier-Stokes equations in divergence form

$$\frac{\partial U}{\partial t} + \nabla \cdot \mathbf{F} = Q \quad U \in \mathbb{R}^5, \quad \mathbf{F} \in \mathbb{R}^{3 \times 5}, \quad Q \in \mathbb{R}^5$$

- representing all equations in the same generic form simplifies the programming
- it suffices to develop discretization techniques for the generic conservation law ▀



Constitutive relations

Variables: $\rho, \mathbf{v}, e, p, \tau, T$

Equations: continuity, momentum, energy



The number of unknowns exceeds the number of equations.

1. Newtonian stress tensor

$$\tau = (\lambda \nabla \cdot \mathbf{v}) \mathcal{I} + 2\mu \mathcal{D}(\mathbf{v}), \quad \mathcal{D}(\mathbf{v}) = \frac{1}{2}(\nabla \mathbf{v} + \nabla \mathbf{v}^T), \quad \lambda \approx -\frac{2}{3}\mu$$

2. Thermodynamic relations, e.g.

$$p = \rho R T \quad \text{ideal gas law}$$

$$R \quad \text{specific gas constant}$$

$$e = c_v T \quad \text{caloric equation of state}$$

$$c_v \quad \text{specific heat at constant volume}$$

Now the system is closed: it contains five PDEs for five independent variables ρ, \mathbf{v}, e and algebraic formulae for the computation of p, τ and T . It remains to specify appropriate initial and boundary conditions.



Initial and Boundary conditions

Initial conditions $\rho|_{t=0} = \rho_0(\mathbf{x}), \quad \mathbf{v}|_{t=0} = \mathbf{v}_0(\mathbf{x}), \quad e|_{t=0} = e_0(\mathbf{x}) \quad \text{in } \Omega$

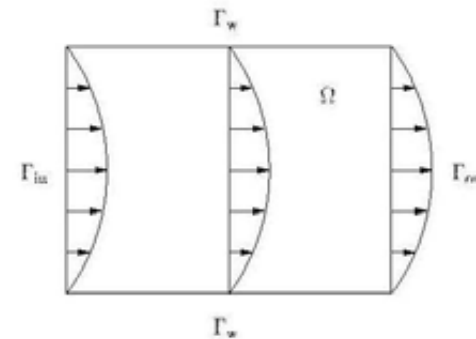
Boundary conditions

Let $\Gamma = \Gamma_{\text{in}} \cup \Gamma_w \cup \Gamma_{\text{out}}$

Inlet $\Gamma_{\text{in}} = \{\mathbf{x} \in \Gamma : \mathbf{v} \cdot \mathbf{n} < 0\}$

$$\rho = \rho_{\text{in}}, \quad \mathbf{v} = \mathbf{v}_{\text{in}}, \quad e = e_{\text{in}}$$

prescribed density, energy and velocity



Solid wall $\Gamma_w = \{\mathbf{x} \in \Gamma : \mathbf{v} \cdot \mathbf{n} = 0\}$

$\mathbf{v} = 0$ no-slip condition

$T = T_w$ given temperature **or**

$\left(\frac{\partial T}{\partial n}\right) = -\frac{f_q}{\kappa}$ prescribed heat flux

Outlet $\Gamma_{\text{out}} = \{\mathbf{x} \in \Gamma : \mathbf{v} \cdot \mathbf{n} > 0\}$

$\mathbf{v} \cdot \mathbf{n} = v_n$ **or** $-p + \mathbf{n} \cdot \boldsymbol{\tau} \cdot \mathbf{n} = 0$

$\mathbf{v} \cdot \mathbf{s} = v_s$ **or** $\mathbf{s} \cdot \boldsymbol{\tau} \cdot \mathbf{n} = 0$

prescribed velocity vanishing stress

The problem is well-posed if the solution exists, is unique and depends continuously on IC and BC. Insufficient or incorrect IC/BC may lead to wrong results (if any).



Dimensionless form of GE

Underlying principle: dynamic similarity of flows

Purpose: equations are normalized in order to

- facilitate the scale-up of obtained results to real flow conditions
- avoid round-off due to manipulations with large/small numbers
- assess the relative importance of terms in the model equations

Dimensionless variables

$$t^* = \frac{t}{t_0}, \quad \mathbf{x}^* = \frac{\mathbf{x}}{L_0}, \quad \mathbf{v}^* = \frac{\mathbf{v}}{v_0}, \quad p^* = \frac{p}{\rho v_0^2}, \quad T^* = \frac{T - T_0}{T_1 - T_0}$$

Dimensionless numbers

Reynolds number $Re = \frac{\rho v_0 L_0}{\mu}$ $\frac{\text{inertia}}{\text{viscosity}}$

Froude number $Fr = \frac{v_0}{\sqrt{L_0 g}}$ $\frac{\text{inertia}}{\text{gravity}}$

Peclet number $Pe = \frac{v_0 L_0}{\kappa}$ $\frac{\text{convection}}{\text{diffusion}}$

Mach number $M = \frac{|\mathbf{v}|}{c}$

Strouhal number $St = \frac{L_0}{v_0 t_0}$

Prandtl number $Pr = \frac{\mu}{\rho \kappa}$ ■



Dimensionless form of GE

Typically:

L : Characteristic Length

$|\mathbf{v}_\infty|$: Inflow/Free-Stream Velocity

ρ_∞ : Inflow/Free-Stream Density

T_∞ : Inflow/Free-Stream Temperature

μ_∞ : Inflow/Free-Stream Viscosity

k_∞ : Inflow/Free-Stream Conductivity



Dimensionless form of GE

Define Non-Dimensional Quantities:

$$t^* = \frac{t |\mathbf{v}_\infty|}{L} , \quad x_i^* = \frac{x_i}{L} , \quad v_i^* = \frac{v_i}{|\mathbf{v}_\infty|}$$

$$\rho^* = \frac{\rho}{\rho_\infty} , \quad T^* = \frac{T}{T_\infty} , \quad p^* = \frac{p}{\rho_\infty |\mathbf{v}_\infty|^2} , \quad e^* = \frac{e}{|\mathbf{v}_\infty|^2}$$

$$\mu^* = \frac{1}{Re_{\infty,L}} \cdot \frac{\mu}{\mu_\infty} , \quad k^* = \frac{1}{(\gamma - 1) M_\infty^2 Pr Re_{\infty,L}} \cdot \frac{k}{k_\infty}$$



Dimensionless form of GE

Characteristic Numbers:

$$Re_{\infty, L} = \frac{\rho_{\infty} |\mathbf{v}_{\infty}| L}{\mu_{\infty}}: \text{Reynolds-Number};$$

Ratio of: Inertial Forces:Viscous Forces

$$M_{\infty} = \frac{|\mathbf{v}_{\infty}|}{c_{\infty}}: \text{Mach-Number};$$

Ratio of: Fluid Velocity:Speed of Sound

$$Pr_{\infty} = \frac{c_p \mu_{\infty}}{k_{\infty}}: \text{Prandtl-Number};$$

Ratio of: Viscosity:Conductivity

$$\gamma = \frac{c_p}{c_v}:$$

Ratio of: Specific Heat at Constant Pressure: _____
Specific Heat at Constant Volume



Dimensionless form of GE

By Going to Dimensionless Form, Dropping the \star :

$$\mathbf{u}_{,t} + \nabla \cdot \mathbf{F}^a = \nabla \cdot \mathbf{F}^v + \mathbf{S}$$

$$\mathbf{u} = \begin{Bmatrix} \rho \\ \rho v_i \\ \rho e \end{Bmatrix}, \quad \mathbf{F}_j^a = \begin{Bmatrix} \rho v_j \\ \rho v_i v_j + p \delta_{ij} \\ v_j(\rho e + p) \end{Bmatrix}, \quad \mathbf{F}_j^v = \begin{Bmatrix} 0 \\ \sigma_{ij} \\ v_l \sigma_{lj} - q_j \end{Bmatrix}$$

$$p = (\gamma - 1)\rho[e - \frac{1}{2}v_j v_j]$$

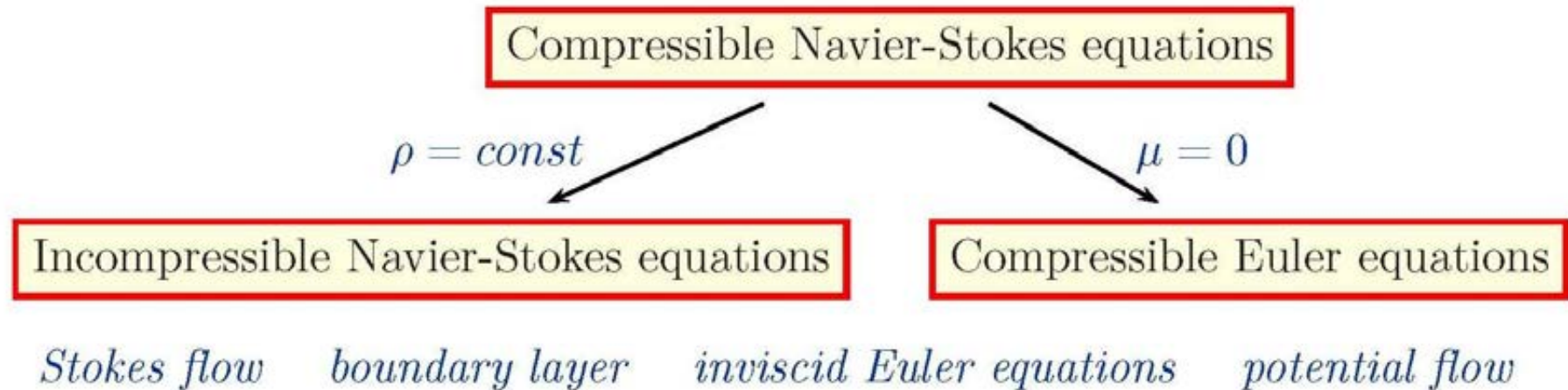
$$\sigma_{ij} = \frac{1}{Re}\mu \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} - \frac{2}{3} \frac{\partial v_k}{\partial x_k} \delta_{ij} \right)$$

$$\mathbf{q} = -\frac{1}{Pr Re} k \nabla T$$



Model Simplification

Purpose: to reduce the computational cost



Derivation of a simplified model

1. determine the type of flow to be simulated
2. separate important and unimportant effects
3. leave irrelevant features out of consideration
4. omit redundant terms/equations from the model
5. prescribe **suitable** initial/boundary conditions



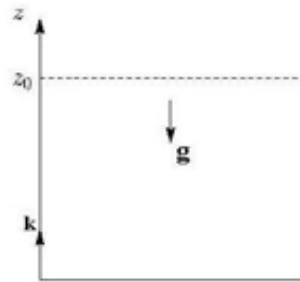
Incompressible flows

Simplification: $\rho = \text{const}$, $\mu = \text{const}$

continuity equation $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0 \longrightarrow \nabla \cdot \mathbf{v} = 0$

inertial term $\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = \rho \left[\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} \right] = \rho \frac{d\mathbf{v}}{dt}$

stress tensor $\nabla \cdot \boldsymbol{\tau} = \mu \nabla \cdot (\nabla \mathbf{v} + \nabla \mathbf{v}^T) = \mu (\nabla \cdot \nabla \mathbf{v} + \nabla \nabla \cdot \mathbf{v}) = \mu \Delta \mathbf{v}$



Let $\rho \mathbf{g} = -\rho g \mathbf{k} = -\nabla(\rho g z) = \nabla p_0$

$p_0 = \rho g(z_0 - z)$ hydrostatic pressure

$\tilde{p} = \frac{p - p_h}{\rho} = \frac{p}{\rho} + g(z_0 - z)$ reduced pressure

$\nu = \frac{\mu}{\rho}$ kinematic viscosity

Incompressible Navier-Stokes equations

$\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} = -\nabla \tilde{p} + \nu \Delta \mathbf{v} - \underbrace{\beta \mathbf{g}(T - T_0)}_{\text{Boussinesq}}$ momentum equations

$\nabla \cdot \mathbf{v} = 0$ continuity equation